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# Modeling Natural Convection From Horizontal Isothermal Annular Heat Sinks

*A study is presented for laminar natural convection heat transfer from isothermal, vertical disks with horizontal support cylinders, as found in annular-fin heat sinks. A distinction is made between the external and internal surfaces of the heat sink, and, hence, the heat transfer mechanisms which dominate in each of these surfaces can be more easily applied. A boundary layer solution for laminar natural convection, a fully developed flow solution, and a diffusive limit are successfully applied to this complex geometry to form the present model. The model shows good agreement with present experimental data and previous correlations.*

## Introduction

Annular-fin heat sinks are used for cooling components in electronic systems and other equipment. A schematic of the annular-fin heat sink used in this study is shown in Fig. 1 in the horizontal orientation, where the gravity force is parallel to the annular fins. The heat sink consists of  $N_f$  identical fins with a centrally located circular support cylinder. Each fin has a fin thickness,  $t$ , an outer diameter,  $D$ , and an inner diameter,  $d$ . Adjacent fins are separated by a distance,  $b$ , and the overall length of the heat sink is  $L$ . The surface of the heat sink was assumed isothermal in this study. Nonisothermal surfaces are also discussed in the Application Related Issues section.

The spacing  $b$  is used as the characteristic length (Elenbaas, 1942; Edwards and Chaddock, 1963; Jones and Nwizu, 1969; Tsubouchi and Masuda, 1970). The Nusselt number is defined as

$$Nu = \frac{Q}{A\Delta T k} b \quad (1)$$

and the Elenbaas Rayleigh number is (Elenbaas, 1942; Edwards and Chaddock, 1963; Jones and Nwizu, 1969; Tsubouchi and Masuda, 1970)

$$Ra = \frac{g\beta\Delta T b^3}{\alpha\nu} \frac{b}{D} \quad (2)$$

where  $\Delta T$  is the temperature difference between the heat sink surface and the ambient air.

As shown in Fig. 1, a heat sink surface consists of four types of component surfaces, i.e., lateral fin surfaces, support cylinder surfaces, fin rim surfaces and end surfaces. The combination of all lateral fin surfaces and support cylinder surfaces is referred to as the inner surface, with the area  $A_{IN}$ ; the fin rim surfaces and the end surfaces are considered as the outer surface, with the area  $A_{OUT}$ . The sum of  $A_{IN}$  and  $A_{OUT}$  is the total heat sink surface area  $A_{HS}$ . The heat transfer mechanism of the inner surface is fundamentally different from that of the outer surface. The inner surface may experience three regimes of heat transfer: thin boundary layer convection at high  $Ra$ , fully developed flow at low  $Ra$  and a transitional regime. While over the outer surface of the heat sink, there is only one mode of heat transfer for the entire range of  $Ra$ , i.e., external natural convection. At low  $Ra$ , the heat flux of the inner surface is much smaller than that of

the outer surface, because the inner surface is in a fully developed regime, while the outer surface is associated with boundary layer heat transfer. Therefore at low  $Ra$ , the surface area ratio  $A_{OUT}/A_{HS}$  is an important factor which determines the magnitude of the heat sink Nusselt number. If  $A_{OUT}/A_{HS}$  is large, the heat sink Nusselt number will be large at low  $Ra$ .

Most previous models for calculating heat transfer from annular heat sinks do not recognize the fundamental difference in heat transfer over the inner and outer surfaces and therefore models based on this premise do not provide an accurate measure of the thermal behavior. Several correlations for natural convection heat transfer from annular-fin heat sinks can be found in the open literature. Edwards and Chaddock (1963) and Jones and Nwizu (1969) did not consider heat transfer from the inner and outer surfaces separately. As a result, their correlations can not be used for heat sinks with different area ratios of  $A_{OUT}/A_{HS}$  from that of their test heat sinks. For example, heat sinks used in electronic systems usually have only a few fins, and the end surface is a large portion of the total surface. The area ratio ( $A_{OUT}/A_{HS}$ ) of these heat sinks are much larger than that of the test heat sinks of Edwards and Chaddock (1963) and Jones and Nwizu (1969), whose test heat sinks had 40 to 70 fins based on the same parameters  $t/D$  and  $b/D$ .

Tsubouchi and Masuda (1970) considered the inner and outer surfaces separately. They gave two correlations for the inner surface and the fin rim surface, respectively. But they did not consider the end surface cooling. They eliminated the heat transfer from the end surfaces in their experiments through the use of end guard sections. As a result their correlations are not suitable for heat sinks used in electronic systems either, where the end surface cooling plays a significant role in the total heat transfer.

The objective of this study is to treat the inner and the outer surfaces separately and establish a general model which accounts for the effects of all surfaces of the heat sink, including the ends and the fin rims. The model will be compared to previous correlations and new experimental data.

## External Natural Convection Solutions

Yovanovich (1987b, c) and Lee et al. (1991) used a characteristic length based on the surface area,  $\sqrt{A}$ , to recast the existing boundary layer solutions for laminar natural convection (Acrivos, 1960; Stewart, 1971; Raithby and Hollands, 1975, 1978) and obtained a new expression of the solutions for isothermal, three dimensional bodies. When combined with the diffusive limit, the expression gives the total heat transfer as

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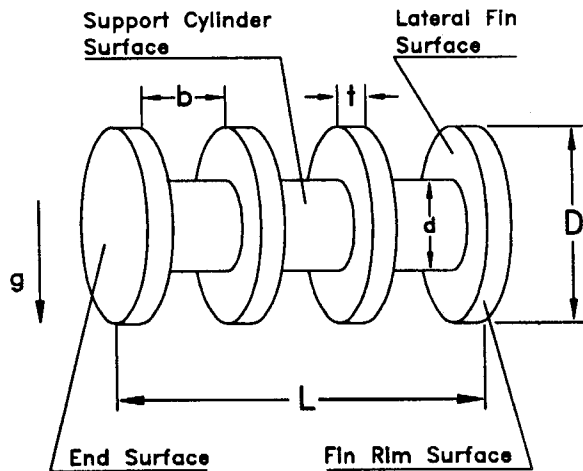


Fig. 1 Dimensions and surfaces of heat sink

$$Nu_{\sqrt{A}} = Nu_{\sqrt{A}}^0 + F(\text{Pr})G_{\sqrt{A}}Ra_{\sqrt{A}}^{1/4} \quad (3)$$

The first term of the equation,  $Nu_{\sqrt{A}}^0$ , is the diffusive limit of the body, which accounts for the correction for the curvature effect of the boundary layers. The second term of the equation,  $F(\text{Pr})G_{\sqrt{A}}Ra_{\sqrt{A}}^{1/4}$ , is the boundary layer solution for the total surface, where  $F(\text{Pr})$  is the approximate "universal" Prandtl number function (Churchill and Churchill, 1975), defined as

$$F(\text{Pr}) = 0.670/[1 + (0.5/\text{Pr})^{9/16}]^{4/9}, \quad (4)$$

and  $G_{\sqrt{A}}$  is the body-gravity function for two-dimensional or axisymmetric surfaces, which is (Yovanovich, 1987b, c)

$$G_{\sqrt{A}} = \left[ \frac{1}{A} \iint_A \left( \frac{P(\theta) \sin \theta}{\sqrt{A}} \right)^{1/3} dA \right]^{3/4}, \quad (5)$$

where  $A$  is the total area of the surface considered,  $P(\theta)$  is the local perimeter of the axisymmetric body, and  $\theta$  is the angle between the outward normal and the gravity vector.

The overall body-gravity function for  $N$  component surfaces connected in parallel with respect to the flow stream can be obtained by (Lee et al., 1991)

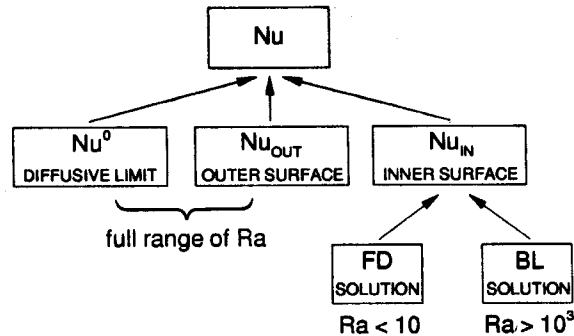


Fig. 2 Model composition

$$G_{\sqrt{A}} = \sum_{i=1}^N G_{\sqrt{A}_i} (A_i/A)^{7/8}, \quad (6)$$

where  $A_i$  refers to the component surface areas and  $A$  is the total surface area for the  $N$  component surfaces, which is given

$$A = \sum_{i=1}^N A_i.$$

The overall body-gravity function for component surfaces connected in series with respect to the flow stream can be obtained from (Lee et al., 1991)

$$G_{\sqrt{A}} = \left[ \sum_{i=1}^N G_{\sqrt{A}_i}^{4/3} (A_i/A)^{7/6} \right]^{3/4}. \quad (7)$$

## Model Development

Figure 2 gives a map of the various model components that will be considered in developing the present model. In Fig. 2,  $Nu$  is the Nusselt number for the heat sink;  $Nu^0$  is the contribution by the diffusive limit;  $Nu_{OUT}$  is the contribution by the convection of the outer surface; and  $Nu_{IN}$  is the contribution by the convection of the inner surface, which is obtained through blending the fully developed flow solution and the boundary layer solution.

Based on the definitions of the Nusselt number, Eq. (1), and the Elenbaas Rayleigh number, Eq. (2), the natural convection solution Eq. (3) can be expressed as

## Nomenclature

$A$  = surface area,  $m^2$   
 $A_{CC}$  = surface area of circumscribed cylinder,  $\equiv \pi D^2/2 + \pi DL$ ,  $m^2$   
 $A_{CS}$  = surface area of channel control surface,  $\equiv \pi Db$ ,  $m^2$   
 $A_{CL}$  = surface area of a single channel,  $\equiv \pi(D^2 - d^2)/2 + \pi db$ ,  $m^2$   
 $A_{IN}$  = inner surface area,  $\equiv (N_f - 1)A_{CL}$ ,  $m^2$   
 $A_{OUT}$  = outer surface area,  $\equiv \pi D^2/2 + N_f \pi D t$ ,  $m^2$   
 $A_{HS}$  = total surface area of heat sink,  $\equiv A_{IN} + A_{OUT}$ ,  $m^2$   
 $b$  = spacing between adjacent fins,  $m$   
 $D$  = fin outer diameter,  $m$   
 $d$  = diameter of support cylinder,  $m$   
 $G_{\sqrt{A}}$  = body-gravity function, Eq. (5)  
 $g$  = gravitational acceleration,  $m/s^2$   
 $k$  = thermal conductivity,  $W/mK$   
 $L$  = length of heat sink,  $\equiv N_f(t + b) - b$ ,  $m$

$N_f$  = total number of fins  
 $Nu^0$  = diffusive limit Nusselt number as  $Ra_D \rightarrow 0$   
 $Nu_{\sqrt{A}}$  = Nusselt number with  $\sqrt{A}$  as the characteristic length,  $\equiv Q\sqrt{A}/A\Delta T k$   
 $Nu$  = Nusselt number with  $b$  as the characteristic length,  $\equiv Qb/A\Delta T k$   
 $n$  = Churchill-Usagi fit parameter  
 $Q$  = heat flow rate by convection,  $W$   
 $Ra_{\sqrt{A}}$  = Rayleigh number with  $\sqrt{A}$  as the characteristic length,  $\equiv g\beta\Delta T(\sqrt{A})^3/\nu\alpha$   
 $Ra$  = Elenbaas Rayleigh number,  $\equiv g\beta\Delta T b^4/\nu\alpha D$   
 $Ra_D$  = Rayleigh number with  $D$  as the characteristic length,  $\equiv g\beta\Delta T D^3/\nu\alpha$   
 $T_s$  = surface temperature of heat sink,  $K$   
 $T_\infty$  = temperature of ambient air,  $K$

$t$  = fin thickness,  $m$   
 $\Delta T$  = temperature difference,  $\equiv T_s - T_\infty$ ,  $K$   
 $\alpha$  = thermal diffusivity of air,  $m^2/s$   
 $\beta$  = volumetric coefficient of thermal expansion,  $1/K$   
 $\nu$  = kinematic viscosity,  $m^2/s$   
**Subscripts**  
 $cf$  = channel flow, based on  $A_{CL}$   
 $cs$  = channel control surface, based on  $A_{CL}$   
 $bl$  = boundary layer flow of inner surface  
 $fd$  = fully developed flow of inner surface  
 $IN$  = inner surface contribution, based on  $A_{HS}$   
 $OUT$  = outer surface contribution, based on  $A_{HS}$

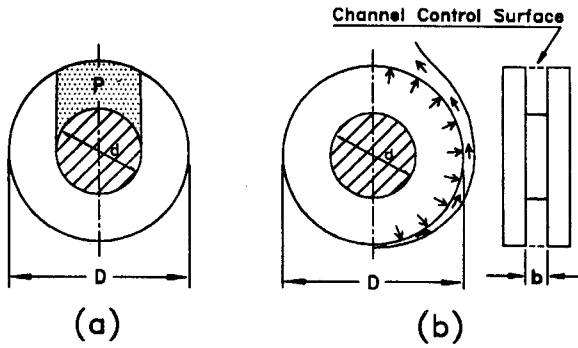


Fig. 3(a) Area affected by cylinder plume; (b) external heat transfer of narrow channel

$$Nu = Nu^0 + F(Pr)G_{\sqrt{A}}(D/\sqrt{A})^{1/4}Ra^{1/4} \quad (8)$$

Equation (8) will be used in the following modeling for air cooling, with  $Pr = 0.71$  and  $F(Pr) = 0.513$ .

**Diffusive Limit.** The diffusive limit of the circumscribed cylinder of the heat sink is taken to represent the heat sink. Based on  $b$  and  $A_{HS}$ , the contribution by the diffusive limit to the total heat transfer of the heat sink can be calculated using (Yovanovich 1987a)

$$Nu^0 = \left\{ \frac{3.1915 + 2.7726 (L/D)^{0.76}}{\sqrt{1 + 2(L/D)}} \right\} \frac{\sqrt{A_{CC}} \cdot b}{A_{HS}} \quad (9)$$

**Nusselt Number for Outer Surface.** It is assumed that the boundary layer flows over the fin rims do not interact with the boundary layer flows associated with the lateral and end surfaces. Because the body-gravity function is insensitive to changes in geometry and orientation of bodies (Lee et al., 1991), the above assumption will not introduce much deviation in the total body-gravity function of the heat sink.

The body-gravity functions of a fin rim and an end surface can be calculated separately using Eq. (5) as  $0.891(t/D)^{1/8}$  and 1.021, respectively. The overall body-gravity function for the outer surface,  $G_{\sqrt{A_{OUT}}}$ , can be obtained using Eq. (6) as

$$G_{\sqrt{A_{OUT}}} = (0.891 t N_f D^{3/4} + 0.607 D^{7/4}) \left( \frac{\pi}{A_{OUT}} \right)^{7/8} \quad (10)$$

The boundary layer heat transfer from the outer surface can be calculated using the second term of Eq. (8). Based on the heat sink surface area  $A_{HS}$ , the contribution of the outer surface to the total heat transfer is

$$Nu_{OUT} = \left\{ 0.513 G_{\sqrt{A_{OUT}}} \left( \frac{D}{\sqrt{A_{OUT}}} \right)^{1/4} Ra^{1/4} \right\} \frac{A_{OUT}}{A_{HS}} \quad (11)$$

### Nusselt Number for Inner Surface.

**Boundary Layer Solution for Inner Surface.** For  $Ra > 10^3$ , the boundary layers formed over adjacent fins do not interact, and, hence, the average heat transfer over the inner surface will be close to that of external convection with the same surface area. In these circumstances, the external natural convection solutions can be directly applied to the inner surface.

For the support cylinder surface between two adjacent fins, the body-gravity function is  $0.891(b/d)^{1/8}$ . In order to take into account the effect of the plume rising from the support cylinder upon the lateral fin surface heat transfer, as shown in Fig. 3(a), the heat transfer rate from area  $P$  will be reduced by half based on the fact that the temperature difference between the area  $P$  and the plume is approximately  $(T_s - T_\infty)/2$ . The body-gravity function of a fin lateral surface, with the effect of the plume,

can be obtained by  $1.021 - 0.317(d/D) + 0.302(d/D)^2$ . The body-gravity functions of the inner surface components are combined using Eq. (6) to give the body-gravity function for the inner surface,  $G_{\sqrt{A_{IN}}}$ .

$$G_{\sqrt{A_{IN}}} = (N_f - 1) \left( \frac{\pi}{A_{IN}} \right)^{7/8} \{ 0.891 b d^{3/4} + [0.607 - 0.188 (d/D) + 0.18 (d/D)^2] (D^2 - d^2)^{7/8} \} \quad (12)$$

The boundary layer Nusselt number for the inner surface can be obtained using the second term of Eq. (8):

$$Nu_{bi} = 0.513 G_{\sqrt{A_{IN}}} (D/\sqrt{A_{IN}})^{1/4} Ra^{1/4} \quad (13)$$

**Fully Developed Flow Solution for Inner Surface.** Two adjacent fins of the heat sink form a channel. In the fully developed regime, the heat transferred from the channel surface consists of the following two parts: (1) the heat carried away by the channel flow; and (2) the heat conducted radially outward from the channel.

1 From the analysis of fully developed flow (Elenbaas, 1942), the Nusselt number for the channel flow based on one channel surface area can be approximated by (Wang, 1997)

$$Nu_{cf} = \frac{1}{12} Ra \frac{D\sqrt{D^2 - d^2}}{A_{CL}} \quad (14)$$

2 When  $Ra$  is very small, e.g., as the spacing  $b$  becomes small, the channel flow will be restricted, and the rate of heat flow carried away by the channel flow will be minimal. Under these circumstances heat will be transferred out of the channel primarily by conduction, and then result in convection outside the channel, as shown in Fig. 3(b). This portion of the heat transfer will be approximately modeled as convective heat transfer from an isothermal, cylindrical surface with diameter  $D$  and width  $b$ . This imaginary surface is referred to as a channel control surface as shown in Fig. 3(b) and its body-gravity function,  $G_{\sqrt{A_{CS}}}$ , is  $0.891(b/D)^{1/8}$ .

The heat transfer at this surface can also be calculated using the second term of Eq. (8) and then converted to be based on the channel surface area  $A_{CL}$ .

$$Nu_{cs} = \left\{ 0.513 G_{\sqrt{A_{CS}}} \left( \frac{D}{\sqrt{A_{CS}}} \right)^{1/4} Ra^{1/4} \right\} \frac{A_{CS}}{A_{CL}} \quad (15)$$

The heat transfer from the inner surface in the fully developed regime can be found by

$$Nu_{fd} = Nu_{cf} + Nu_{cs} \quad (16)$$

**Blending Two Limiting Solutions.** The composite solution technique suggested by Churchill and Usagi (1972) is used to combine the two limiting solutions to give the inner surface Nusselt number for a full range of  $Ra$ . Based on  $A_{HS}$ , the inner surface contribution to the total heat transfer is

$$Nu_{IN} = \frac{A_{IN}/A_{HS}}{[(1/Nu_{bi})^n + (1/Nu_{fd})^n]^{1/n}} \quad (17)$$

The concept of combining the two limits to give the channel solution first appeared in the work of Churchill (1977). Based on the comparisons with the present experimental data and the previous correlations,  $n = 1$  was chosen for heat sink applications.

This value of  $n$  is different from the previous values of  $n$  for parallel plates, such as  $n = 1.5$  (Churchill, 1977),  $n = 2$  (Kraus and Bar-Cohen, 1983),  $n = 1.9$  (Raithby and Hollands, 1985), because the interference of the support cylinders changes the behavior of the channel heat transfer. Actually if different values of  $n$  are used for different diameter ratios of  $D/d$ , the model will give better predictions for each case of  $D/d$ . But for the

**Table 1 Functions of simplified model**

Functions	Parameter Ranges
$f_1 = [3.36 + 0.087(L/D)]\sqrt{Acc} \cdot b/A_{HS}$	$L/D < 10$
$f_2 = [0.449 - 0.026 \ln(t N_f/D)] A_{OUT}/A_{HS}$	$0.01 < t N_f/D < 10$
$f_3 = 0.573 - 0.184(d/D) + 0.0388(d/D)^2$	$6.3 \times 10^5 < Ra_D < 1.2 \times 10^6$ and $0.1 \leq d/D \leq 0.8$
$f_4 = 0.0516 + 0.0154(d/D) - 0.0433(d/D)^2 + 0.0792(d/D)^3$	
$f_5 = 0.0323 - 0.0517(d/D) + 0.11(d/D)^2$	

purpose of simplicity,  $n = 1$  was chosen for heat sink applications. In practice the diameter ratio  $D/d$  of the heat sinks does not change very much, usually ranging from 1.6 to 3.6 for the heat sinks used in electronic systems.

**Nusselt Number for Heat Sink.** The total heat transfer from the heat sink can be found by combining the three contributions, namely, the heat transfer from the inner surface, the boundary layer heat transfer from the outer surface, and the diffusive limit. Therefore, the model for the heat sink is obtained as

$$Nu = Nu^0 + Nu_{OUT} + Nu_{IN} \quad (18)$$

**Simplified Model.** In order to make the calculation of  $Nu$  in Eq. (18) easier, the following correlation equations were obtained through fitting the predictions of Eq. (9), Eq. (11), and Eqs. (13–15):

$$Nu^0 = f_1 \quad (19)$$

$$Nu_{OUT} = f_2 Ra^{1/4} \quad (20)$$

$$Nu_{IN} = \frac{A_{IN}/A_{HS}}{1/Nu_{bi} + 1/Nu_{fd}} \quad (21)$$

where

$$Nu_{bi} = f_3 Ra^{1/4} \quad (22)$$

$$Nu_{fd} = f_4 Ra + f_5 Ra^{1/4} \quad (23)$$

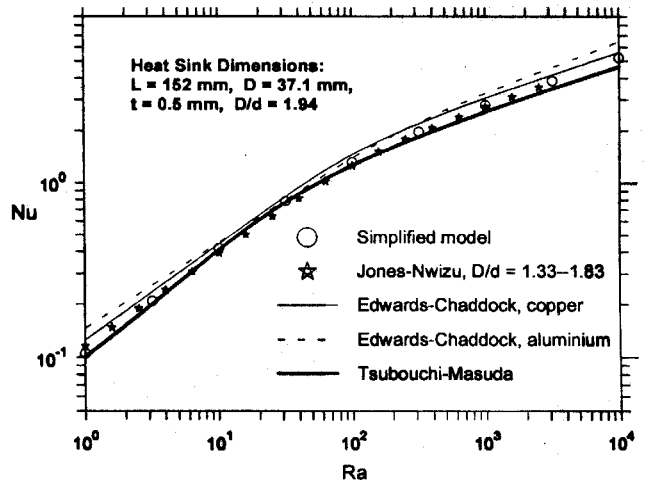
The functions  $f_1, f_2, f_3, f_4, f_5$  and their parameter ranges are given in Table 1. For the range of  $Ra$  from 10 to  $10^4$ , the maximum difference between the predictions by the model and its simplified equations is within 5 percent and the RMS difference is within 3 percent. Although the model was simplified for heat sinks within the parameter ranges, it may also be used outside the ranges.

### Trends of Model Predictions

The model predictions for the heat sinks used in this study are shown in Fig. 6 for a wide range of  $Ra$  between  $10^{-2}$  and  $10^5$ . They are the five fine curves A, B, C, D, and E. The geometric dimensions of the five heat sinks are given in Table 2. In these plots, geometric parameters are fixed for each curve,  $Ra$  varies with  $Ra_D$  (note:  $Ra = Ra_D \cdot b^4/D^4$ ). The top dashed line represents a solid horizontal circular cylinder, (specifically,

**Table 2 Geometric parameters of test heat sinks ( $d = 22$ ,  $D/d = 1.66$ , length unit: mm)**

Heat Sink	$b$	$t$	$N_f$	$L$	$D$	$A_{HS}$
A	22.5	10	3	75	36.5	11308
B	7.5	9	5	75	36.5	14656
C	4	9	6	74	36.5	16329
D	2	9	7	75	36.5	18141
E	1	10	7	76	36.5	18529



**Fig. 4 Comparisons of simplified model with previous correlations ( $D/d = 1.94$ )**

$L = 75$  mm,  $D = 36.5$  mm,  $d/D = 0.999$ ,  $t = 20$  mm,  $b = 35$  mm, and  $N_f = 2$ ). The bottom thick solid curve represents a heat sink with the same  $d/D$  as the test heat sinks but with very small fin thickness and a large number of fins. In this case, the area ratio  $A_{OUT}/A_{HS}$  is very small, the inner surface heat transfer is dominant in the entire range of  $Ra$ , so the curve is similar to a channel heat transfer curve. For the heat sinks, when  $b/D$  is large, there are fewer fins and the area ratio  $A_{OUT}/A_{HS}$  is relatively large, the channeling effect will be small and the heat transfer curve will be similar to a solid cylinder curve, i.e., curve A in Fig. 6. When  $b/D$  is small, especially with a small fin thickness  $t$  and a large number of fins,  $N_f$ , the area ratio  $A_{OUT}/A_{HS}$  is small, the channeling effect will be large and the heat transfer curve will be close to the channel curve.

If  $L/D$ ,  $t/D$ , and  $d/D$  of the heat sink and  $Pr$  are fixed, as is common practice by most researchers,  $Nu$  is a function of two parameters, i.e.,  $Nu = f(Ra, b/D)$ , as shown in Fig. 6, where for each value of  $Ra$  there are multiple values of  $Nu$  corresponding to different heat sinks or different  $b/D$ . Similar trends of numerical results for parallel plate channels were reported previously by Martin et al. (1991) and Li and Chung (1996). Therefore, comparisons with previous experimental data should be made in such a way that at each  $Ra$  value, the present value of  $b/D$  is the same as that in the previous experiments, especially at low  $Ra$ . And the end surfaces and the fin thickness should also be taken into consideration.

### Comparisons to Previous Correlations and Present Experimental Data

In the previous experiments (Edwards and Chaddock, 1963; Jones and Nwizu, 1969; Tsubouchi and Masuda, 1970),  $Ra_D$  was more or less fixed and the wide range of  $Ra$  was achieved by variation of  $b/D$  (note:  $Ra = Ra_D \cdot b^4/D^4$ ). Therefore, in the following comparisons, the present model predictions are given in such a way that  $Ra_D$  is fixed at an average value and  $Ra$  varies with  $b/D$ . In addition, the heat sink dimensions used by Edwards and Chaddock (1963) are adopted for the present model. In their experiments, the length of the heat sinks,  $L$ , the thickness of fins,  $t$ , and the diameter of support cylinder,  $d$ , were fixed. They made measurements for three cases of the diameter ratio ( $D/d = 1.94, 2.97$ , and  $5.17$ ).

In the previous experiments, the radiation losses were calculated and subtracted from the total heat flow rate.

In Fig. 4, the simplified model is compared with the previous correlations for  $D/d = 1.94$ . The maximum difference between the model and the Jones-Nwizu correlation is 9.8 percent and

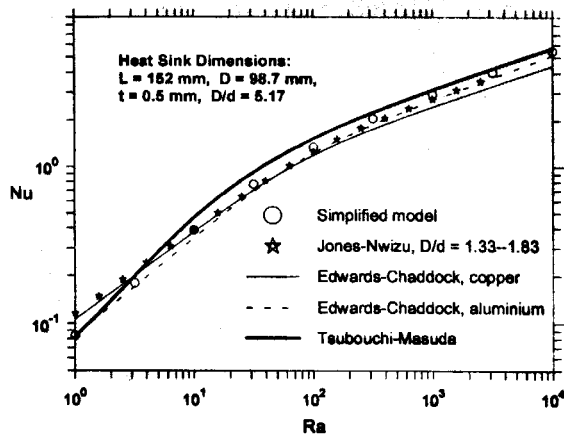


Fig. 5 Comparisons of simplified model with previous correlations ( $D/d = 5.17$ )

the RMS difference is 5.4 percent. While the maximum and the RMS differences between the model and the Tsubouchi-Masuda correlation are 11.3 percent and 6.4 percent, respectively.

The comparisons for  $D/d = 5.17$  are shown in Fig. 5. Between the model and the Edwards-Chaddock correlation for aluminum fins, the maximum and RMS differences are 11.5 percent and 8.1 percent, respectively. From  $Ra = 10$  to  $10^4$ , the maximum difference between the model and the Jones-Nwizu correlation is 7.1 percent and the RMS difference is 6.2 percent.

The present model takes into account the end surface heat transfer, while the Jones-Nwizu and the Tsubouchi-Masuda correlations do not. Because the number of fins is quite large in the above comparisons ( $N_f = 80$  for  $Ra = 1$ ), the end surface effect becomes negligible.

Five heat sinks were tested in the present experiments. Their geometric dimensions are given in Table 2, where  $N_f$  is the number of fins and  $A_{HS}$  is the total surface area of the heat sink. A cartridge heater was embedded in the center of the heat sink to provide a source for internal heat generation. The heat sink was suspended horizontally, as shown in Fig. 1, in the center of a large enclosure with quiescent air as the cooling fluid. All surfaces, including the two end surfaces, were exposed to the cooling fluid. The experiments were carried out at atmospheric pressure, and the temperature differences between the heat sink and the ambient air ranged from 20 K to 80 K. Six to eight thermocouples were embedded under the surface of the heat sink at various locations to provide a true measure of the heat sink temperature. Care was taken to prevent the thermocouple wire heat transfer from disturbing the temperature fields around the thermocouples. The maximum temperature difference between any two thermocouples was 0.25 K when  $T_s - T_\infty = 50$  K, so the heat sink surface was considered as isothermal.

The radiation losses were measured in a vacuum chamber for each heat sink at different temperature levels. In data reduction, these amounts of radiation losses were subtracted from the total heat transfer rate measured in convection experiment. The heat losses through the power leads and the thermocouple wires were also accounted for in the data reduction. The uncertainty analysis (Wang, 1997) showed that the uncertainty in  $Ra$  was within  $\pm 5.1$  percent and the uncertainty in  $Nu$  was within  $\pm 2.0$  percent.

The experimental results are compared with the simplified model in Fig. 6 along with three other correlations, where the dimensions of the test heat sinks are used in the model. In this comparison  $Ra$  varies with  $Ra_D$ . Although the five heat sinks were tested at the similar conditions, i.e., the  $Ra_D$  are close to one another, the experimental results (square symbols) lie in different ranges of  $Ra$  because they have different values of  $b/$

$D$ . In the order  $A, B, C, D$ , and  $E$ , the experimental results of the heat sinks lie in the range of  $Ra = 2 \times 10^4$  to  $10^{-1}$ . The curves  $A, B, C, D$ , and  $E$  represent the model predictions for the five heat sinks. The agreement between the experimental data and the model is very good. The maximum and RMS percentage differences between the model and the data for each of the five heat sinks are  $A(7.5, 5.1)$ ,  $B(6.4, 5.6)$ ,  $C(4.2, 2.8)$ ,  $D(3.3, 2.5)$ , and  $E(3.5, 2.5)$ , respectively.

In the range of  $Ra > 4 \times 10^3$  the boundary layer heat transfer is dominant, so the experimental results of heat sink  $A$  lie along a straight line of slope  $\frac{1}{4}$ . For  $Ra = 10^3$  to  $10^0$ , the introduction of channeling effects results in increases in the slopes of the curves. For  $Ra < 10^{-1}$ , the boundary layer heat transfer from the outer surface of heat sink  $E$  becomes dominant, and the slope of curve  $E$  returns to  $\frac{1}{4}$ . Figure 6 also shows that the previous correlations can not predict heat transfer from these heat sinks at low  $Ra$ . Since there are just a few fins on these heat sinks, the end surfaces play a significant role in the total heat transfer, and the heat transfer from the fin rim surfaces is significant in these cases due to the large fin thickness. These two factors were not considered in the previous correlations, therefore they cannot account for the heat transfer from the end surfaces and the larger fin rim surfaces, and, hence, give lower predictions for the test heat sinks at low  $Ra$ .

## Application Related Issues

**Radiative Dissipation.** In the natural convection regime, radiation can play an important role in electronic cooling. A set of equations for calculating radiative losses from annular heat sinks was given by Wang et al. (1997) and Wang (1997).

**Mounted Heat Sinks.** Annular-fin heat sinks used in electronic applications are usually mounted by bonding one end of the heat sink to a flat surface of a component, while the other end is exposed to the cooling fluid. Therefore, only one end of the heat sink acts as an active surface for natural convection cooling. Because the proposed method is flexible, the boundary condition on each of the end surfaces can be specified independently, allowing this configuration to be modeled.

**Nonisothermal Surface.** Heat sinks used in practical applications are not truly isothermal, although in many cases the temperature variation on the heat sink surface is small. To account for this temperature variation, an average temperature of the heat sink surface may need to be found before using the model for the heat sinks. This can be done through a fin efficiency analysis. The average temperature can be taken as the isothermal temperature of the heat sink in the model. Hahne and Zhu (1994) made comparisons between two sets of results,

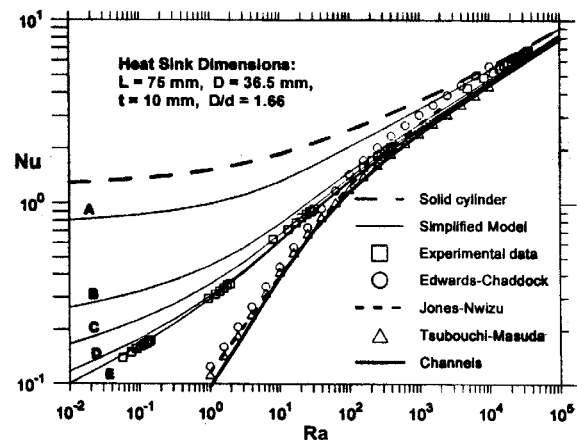


Fig. 6 Comparison of simplified model with present experimental data

one was obtained using the total surface area and the average temperature, the other was calculated with subsurface areas and their local temperatures. They found that the difference between the two sets of data was less than 1 percent. This supports the approach stated above. In cases with small surface temperature variations, finding the average surface temperature may not be necessary and the isothermal assumption will provide accurate results.

## Summary and Conclusions

The boundary layer solution for laminar natural convection, the fully developed flow solution and the diffusive limit are successfully applied to the isothermal vertical disks with horizontal support cylinders. A general model for laminar natural convection heat transfer from this geometry has been developed and applied to annular heat sinks. The model accounts for the effects of the end surfaces and the fin thickness, and overcomes the shortcomings of the previous correlations.

A simplified model is presented and compared with previous correlations and new experimental data. Very good agreement can be seen in these comparisons. The previous correlations fail to predict the present experimental data at the fully developed regime, because they do not take into consideration the effects of the end surfaces and the fin thickness.

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