15-18 n-Octane is burned with stoichiometric amount of air. The mass fraction of each product, the mass of water in the products and the mass fraction of each reactant are to be determined.

Assumptions 1 Combustion is complete. 2 The combustion products contain CO<sub>2</sub>, H<sub>2</sub>O, O<sub>2</sub>, and N<sub>2</sub> only.

Properties The molar masses of C, H<sub>2</sub>, O<sub>2</sub> and air are 12 kg/kmol, 2 kg/kmol, 32 kg/kmol, and 29 kg/kmol, respectively (Table A-1).

Analysis The reaction equation for 100% theoretical air is

$$C_8H_{18} + a_{th}[O_2 + 3.76N_2] \longrightarrow BCO_2 + DH_2O + EN_2$$

where  $a_{\rm th}$  is the stoichiometric coefficient for air. The coefficient  $a_{\rm th}$ and other coefficients are to be determined from the mass balances

**Products** Combustion chamber 100% theoretical

 $C_4H_{10}$ 

B = 8Carbon balance:

$$2D = 18 \longrightarrow D = 9$$

Hydrogen balance: 
$$2D = 18 \longrightarrow D = 9$$

Oxygen balance: 
$$2a_{th} = 2B + D \longrightarrow a_{th} = 0.5(2 \times 8 + 9) = 12.5$$

Nitrogen balance: 
$$a_{th} \times 3.76 = E \longrightarrow E = 12.5 \times 3.76 = 47$$

Substituting, the balanced reaction equation is

$$C_8H_{18} + 12.5[O_2 + 3.76N_2] \longrightarrow 8CO_2 + 9H_2O + 47N_2$$

The mass of each product and the total mass are

$$\begin{split} m_{\rm CO2} &= N_{\rm CO2} M_{\rm CO2} = (8\,{\rm kmol})(44\,{\rm kg/kmol}) = 352\,{\rm kg} \\ m_{\rm H2O} &= N_{\rm H2O} M_{\rm H2O} = (9\,{\rm kmol})(18\,{\rm kg/kmol}) = 162\,{\rm kg} \\ m_{\rm N2} &= N_{\rm N2} M_{\rm N2} = (47\,{\rm kmol})(28\,{\rm kg/kmol}) = 1316\,{\rm kg} \\ m_{\rm total} &= m_{\rm CO2} + m_{\rm N2} + m_{\rm H2O} = 352 + 162 + 1316 = 1830\,{\rm kg} \end{split}$$

Then the mass fractions are

$$mf_{CO2} = \frac{m_{CO2}}{m_{total}} = \frac{352 \text{ kg}}{1830 \text{ kg}} = \mathbf{0.1923}$$

$$mf_{H2O} = \frac{m_{H2O}}{m_{total}} = \frac{162 \text{ kg}}{1830 \text{ kg}} = \mathbf{0.0885}$$

$$mf_{N2} = \frac{m_{N2}}{m_{total}} = \frac{1316 \text{ kg}}{1830 \text{ kg}} = \mathbf{0.7191}$$

The mass of water per unit mass of fuel burned is

$$\frac{m_{\rm H2O}}{m_{\rm C8H18}} = \frac{(9 \times 18) \,\text{kg}}{(1 \times 114) \,\text{kg}} = \textbf{1.421 kg} \, \textbf{H}_{\textbf{2}} \textbf{O/kg} \, \textbf{C}_{\textbf{8}} \textbf{H}_{\textbf{18}}$$

The mass of each reactant and the total mass are

$$m_{\text{C8H18}} = N_{\text{C8H18}} M_{\text{C8H18}} = (1 \text{ kmol})(114 \text{ kg/kmol}) = 114 \text{ kg}$$
  
 $m_{\text{air}} = N_{\text{air}} M_{\text{air}} = (12.5 \times 4.76 \text{ kmol})(29 \text{ kg/kmol}) = 1725.5 \text{ kg}$   
 $m_{\text{total}} = m_{\text{C8H18}} + m_{\text{air}} = 114 + 1725.5 = 1839.5 \text{ kg}$ 

Then the mass fractions of reactants are

$$mf_{C8H18} = \frac{m_{C8H18}}{m_{total}} = \frac{114 \text{ kg}}{1839.5 \text{ kg}} = \mathbf{0.0620}$$
$$mf_{air} = \frac{m_{air}}{m_{total}} = \frac{1725.5 \text{ kg}}{1839.5 \text{ kg}} = \mathbf{0.9380}$$

**15-27** Butane  $C_4H_{10}$  is burned with 200 percent theoretical air. The kmol of water that needs to be sprayed into the combustion chamber per kmol of fuel is to be determined.

Assumptions 1 Combustion is complete. 2 The combustion products contain CO<sub>2</sub>, H<sub>2</sub>O, O<sub>2</sub>, and N<sub>2</sub> only.

**Properties** The molar masses of C, H<sub>2</sub>, O<sub>2</sub> and air are 12 kg/kmol, 2 kg/kmol, 32 kg/kmol, and 29 kg/kmol, respectively (Table A-1).

Analysis The reaction equation for 200% theoretical air without the additional water is

$$C_4H_{10} + 2a_{th}[O_2 + 3.76N_2] \longrightarrow BCO_2 + DH_2O + EO_2 + FN_2$$

where  $a_{th}$  is the stoichiometric coefficient for air. We have automatically accounted for the 100% excess air by using the factor  $2a_{th}$  instead of  $a_{th}$  for air. The coefficient  $a_{th}$  and other coefficients are to be determined from the mass balances

Carbon balance: B = 4

Hydrogen balance:  $2D = 10 \longrightarrow D = 5$ Oxygen balance:  $2 \times 2a_{th} = 2B + D + 2E$ 

 $a_{th} = E$ 

Air 200% theoretical

 $C_4H_{10}$ 

Nitrogen balance:

$$2a_{th} \times 3.76 = F$$

Solving the above equations, we find the coefficients (E = 6.5, F = 48.88, and  $a_{th} = 6.5$ ) and write the balanced reaction equation as

$$C_4H_{10} + 13[O_2 + 3.76N_2] \longrightarrow 4CO_2 + 5H_2O + 6.5O_2 + 48.88N_2$$

With the additional water sprayed into the combustion chamber, the balanced reaction equation is

$$C_4H_{10} + 13[O_2 + 3.76N_2] + N_v H_2O \longrightarrow 4CO_2 + (5 + N_v) H_2O + 6.5 O_2 + 48.88 N_2$$

The partial pressure of water in the saturated product mixture at the dew point is

$$P_{v \text{ prod}} = P_{\text{sat} @ 60^{\circ}\text{C}} = 19.95 \text{ kPa}$$

The vapor mole fraction is

$$y_v = \frac{P_{v,\text{prod}}}{P_{\text{prod}}} = \frac{19.95 \text{ kPa}}{100 \text{ kPa}} = 0.1995$$

The amount of water that needs to be sprayed into the combustion chamber can be determined from

$$y_v = \frac{N_{\text{water}}}{N_{\text{total product}}} \longrightarrow 0.1995 = \frac{5 + N_v}{4 + 5 + N_v + 6.5 + 48.88} \longrightarrow N_v =$$
**9.796 kmol**

15-51 The higher and lower heating values of liquid propane are to be determined and compared to the listed values.

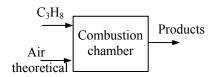
Assumptions 1 Combustion is complete. 2 The combustion products contain CO<sub>2</sub>, H<sub>2</sub>O, and N<sub>2</sub>. 3 Combustion gases are ideal gases.

Properties The molar masses of C, O<sub>2</sub>, H<sub>2</sub>, and air are 12, 32, 2, and 29 kg/kmol, respectively (Table A-1).

Analysis The combustion reaction with stoichiometric air is

$$C_3H_8(l) + 5(O_2 + 3.76N_2) \longrightarrow 3CO_2 + 4H_2O + 18.8N_2$$

Both the reactants and the products are taken to be at the standard reference state of 25°C and 1 atm for the calculation of heating values. The heat transfer for this process is equal to enthalpy of combustion. Note that  $N_2$  and  $O_2$  are stable elements, and thus their enthalpy of formation is zero. Then,



$$q = h_C = H_P - H_R = \sum_{p} N_p \overline{h}_{f,p}^{\circ} - \sum_{p} N_p \overline{h}_{f,R}^{\circ} = \left(N \overline{h}_f^{\circ}\right)_{CO2} + \left(N \overline{h}_f^{\circ}\right)_{H2O} - \left(N \overline{h}_f^{\circ}\right)_{C3H8}$$

The  $\overline{h}_f^{\circ}$  of liquid propane is obtained by adding  $\overline{h}_{fg}$  of propane at 25°C to  $\overline{h}_f^{\circ}$  of gas propane (103,850 + 44.097 × 335 = 118,620 kJ/kmol). For the HHV, the water in the products is taken to be liquid. Then,

$$h_C = (3 \text{ kmol})(-393,520 \text{ kJ/kmol}) + (4 \text{ kmol})(-285,830 \text{ kJ/kmol}) - (1 \text{ kmol})(-118,620 \text{ kJ/kmol})$$
  
= -2,205,260 kJ/kmol propane

The HHV of the liquid propane is

$${\rm HHV} = \frac{-h_C}{M_m} = \frac{2{,}205{,}260~{\rm kJ/kmol}~{\rm C_3H_8}}{44.097~{\rm kg/kmol}~{\rm C_3H_8}} = \textbf{50,010~kJ/kg}~{\rm \textbf{C_3H_8}}$$

The listed value from Table A-27 is **50,330 kJ/kg**. For the LHV, the water in the products is taken to be vapor. Then,

$$h_C = (3 \text{ kmol})(-393,520 \text{ kJ/kmol}) + (4 \text{ kmol})(-241,820 \text{ kJ/kmol}) - (1 \text{ kmol})(-118,620 \text{ kJ/kmol})$$
  
= -2,029,220 kJ/kmol propane

The LHV of the propane is then

LHV = 
$$\frac{-h_C}{M_m}$$
 =  $\frac{2,029,220 \text{ kJ/kmol C}_3\text{H}_8}{44.097 \text{ kg/kmol C}_3\text{H}_8}$  = **46,020 kJ/kg C**<sub>3</sub>H<sub>8</sub>

The listed value from Table A-27 is **46,340 kJ/kg**. The calculated and listed values are practically identical.

**15-66** A mixture of propane and methane is burned with theoretical air. The balanced chemical reaction is to be written, and the amount of water vapor condensed and the the required air flow rate for a given heat transfer rate are to be determined.

**Assumptions 1** Combustion is complete. **2** The combustion products contain CO<sub>2</sub>, CO, H<sub>2</sub>O, O<sub>2</sub>, and N<sub>2</sub> only.

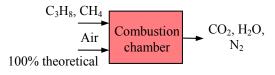
**Properties** The molar masses of C, H<sub>2</sub>, O<sub>2</sub>, N<sub>2</sub> and air are 12 kg/kmol, 2 kg/kmol, 32 kg/kmol, 28 kg/kmol, and 29 kg/kmol, respectively (Table A-1).

Analysis (a) The balanced reaction equation for stoichiometric air is

$$0.4 \,\mathrm{C_3H_8} + 0.6 \,\mathrm{CH_4} + a_{\mathrm{th}} \left[\mathrm{O_2} + 3.76 \,\mathrm{N_2}\right] \longrightarrow 1.8 \,\mathrm{CO_2} + 2.8 \,\mathrm{H_2O} + a_{\mathrm{th}} \times 3.76 \,\mathrm{N_2}$$

The stoicihiometric coefficient  $a_{th}$  is determined from an  $O_2$  balance:

$$a_{\rm th} = 1.8 + 1.4 = 3.2$$



Substituting,

$$0.4 C_3 H_8 + 0.6 CH_4 + 3.2 [O_2 + 3.76 N_2] \longrightarrow 1.8 CO_2 + 2.8 H_2 O + 12.032 N_2$$

(b) The partial pressure of water vapor is

$$P_{v} = \frac{N_{\text{H2O}}}{N_{\text{total}}} P_{\text{total}} = \frac{2.8}{1.8 + 2.8 + 12.032} (100 \text{ kPa}) = \frac{2.8 \text{ kmol}}{16.632 \text{ kmol}} (100 \text{ kPa}) = 16.84 \text{ kPa}$$

The dew point temperature of the product gases is the saturation temperature of water at this pressure:

$$T_{dp} = T_{\text{sat}@16.84 \text{ kPa}} = 56.2^{\circ}\text{C}$$
 (Table A-5)

Since the temperature of the product gases are at 398 K (125°C), there will be no condensation of water vapor.

(c) The heat transfer for this combustion process is determined from the energy balance  $E_{\rm in} - E_{\rm out} = \Delta E_{\rm system}$  applied on the combustion chamber with W = 0. It reduces to

$$-Q_{\rm out} = \sum N_P \left(\overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ}\right)_P - \sum N_R \left(\overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ}\right)_R$$

The products are at 125 °C, and the enthalpy of products can be expressed as

$$(\overline{h} - \overline{h}^{\circ}) = \overline{c}_p \Delta T$$

where  $\Delta T = 125 - 25 = 100$  °C = 100 K . Then, using the values given in the table,

$$-Q_{\text{out}} = (1.8)(-393,520 + 41.16 \times 100) + (2.8)(-241,820 + 34.28 \times 100) + (12.032)(0 + 29.27 \times 100)$$
$$- (0.4)(-103,850) - (0.6)(-74,850)$$
$$= -1,246,760 \text{ kJ/kmol fuel}$$

or  $Q_{out} = 1,246,760 \text{ kJ/kmol fuel}$ 

For a heat transfer rate of 97,000 kJ/h, the molar flow rate of fuel is

$$\dot{N}_{\text{fuel}} = \frac{\dot{Q}_{\text{out}}}{Q_{\text{out}}} = \frac{97,000 \text{ kJ/h}}{1,246,760 \text{ kJ/kmol fuel}} = 0.07780 \text{ kmol fuel/h}$$

The molar mass of the fuel mixture is

$$M_{\text{fuel}} = 0.4 \times 44 + 0.6 \times 16 = 27.2 \text{ kg/kmol}$$

The mass flow rate of fuel is

$$\dot{m}_{\text{fuel}} = \dot{N}_{\text{fuel}} M_{\text{fuel}} = (0.07780 \,\text{kmol/h})(27.2 \,\text{kg/kmol}) = 2.116 \,\text{kg/h}$$

The air-fuel ratio is

AF = 
$$\frac{m_{\text{air}}}{m_{\text{fuel}}} = \frac{(3.2 \times 4.76 \times 29) \text{ kg}}{(0.4 \times 44 + 0.6 \times 16) \text{ kg}} = 16.24 \text{ kg air/kg fuel}$$

The mass flow rate of air is then

$$\dot{m}_{\rm air} = \dot{m}_{\rm fuel} AF = (2.116 \,\text{kg/h})16.24) = 34.4 \,\text{kg/h}$$

Combustion

chamber

**Products** 

15-73 A high efficiency gas furnace burns gaseous propane  $C_3H_8$  with 140 percent theoretical air. The volume flow rate of water condensed from the product gases is to be determined.

Assumptions 1 Combustion is complete. 2 The combustion products contain CO<sub>2</sub>, H<sub>2</sub>O, O<sub>2</sub>, and N<sub>2</sub> only.

Properties The molar masses of C, H<sub>2</sub>, O<sub>2</sub> and air are 12 kg/kmol, 2 kg/kmol, 32 kg/kmol, and 29 kg/kmol, respectively (Table A-1).

Analysis The reaction equation for 40% excess air (140% theoretical air) is

$$C_3H_8 + 1.4a_{th}[O_2 + 3.76N_2] \longrightarrow BCO_2 + DH_2O + EO_2 + FN_2$$

where  $a_{th}$  is the stoichiometric coefficient for air. We have automatically accounted for the 40% excess air by using the factor 1.4 $a_{th}$  instead of  $a_{th}$  for air. The coefficient  $a_{th}$  and other coefficients are to be determined from the mass balances

Carbon balance:

 $2D = 8 \longrightarrow D = 4$ Hydrogen balance:

Oxygen balance:  $2 \times 1.4 a_{th} = 2B + D + 2E$ 

 $0.4a_{\rm th} = E$ 

 $1.4a_{th} \times 3.76 = F$ Nitrogen balance:

Solving the above equations, we find the coefficients (E = 2,

F = 26.32, and  $a_{th} = 5$ ) and write the balanced reaction equation as

$$C_3H_8 + 7[O_2 + 3.76N_2] \longrightarrow 3CO_2 + 4H_2O + 2O_2 + 26.32N_2$$

The partial pressure of water in the saturated product mixture at the dew point is

$$P_{v,\text{prod}} = P_{\text{sat@}40^{\circ}\text{C}} = 7.3851 \text{ kPa}$$

The vapor mole fraction is

$$y_v = \frac{P_{v,\text{prod}}}{P_{\text{prod}}} = \frac{7.3851 \,\text{kPa}}{100 \,\text{kPa}} = 0.07385$$

The kmoles of water condensed is determined from

$$y_v = \frac{N_{\text{water}}}{N_{\text{total, product}}} \longrightarrow 0.07385 = \frac{4 - N_{\text{w}}}{3 + 4 - N_{\text{w}} + 2 + 26.32} \longrightarrow N_{\text{w}} = 1.503 \text{ kmol}$$

The steady-flow energy balance is expressed as

$$\dot{N}_{\text{fuel}} H_R = \dot{Q}_{\text{fuel}} + \dot{N}_{\text{fuel}} H_P$$

where 
$$\dot{Q}_{\text{fuel}} = \frac{\dot{Q}_{\text{out}}}{\eta_{\text{furnace}}} = \frac{31,650 \text{ kJ/h}}{0.96} = 32,969 \text{ kJ/h}$$

$$H_R = \overline{h}_{f \text{ fuel@25°C}}^o + 7\overline{h}_{O2@25^{\circ}C} + 26.32\overline{h}_{N2@25^{\circ}C}$$

= 
$$(-103,847 \text{ kJ/kmol}) + 7(0) + 26.32(0) = -103,847 \text{ kJ/kmol}$$

$$H_P = 3\overline{h}_{\text{CO2}@25^{\circ}\text{C}} + 4\overline{h}_{\text{H2O}@25^{\circ}\text{C}} + 2\overline{h}_{\text{O2}@25^{\circ}\text{C}} + 26.32\overline{h}_{\text{N2}@25^{\circ}\text{C}} + N_{\text{w}}(\overline{h}_{f}^{\,o}_{\text{H2O(liq)}})$$

$$= 3(-393,\!520\,\mathrm{kJ/kmol}) + 4(-241,\!820\,\mathrm{kJ/kmol}) + 2(0) + 26.32(0) + 1.503(-285,\!830\,\mathrm{kJ/kmol})$$

$$= -2.577 \times 10^6 \text{ kJ/kmol}$$

Substituting into the energy balance equation,

$$\dot{N}_{\text{fuel}} H_R = \dot{Q}_{\text{fuel}} + \dot{N}_{\text{fuel}} H_P$$

$$\dot{N}_{\rm fuel}$$
 (-103,847 kJ/kmol) = 32,969 kJ/h +  $\dot{N}_{\rm fuel}$  (-2.577×10<sup>6</sup> kJ/kmol)  $\longrightarrow \dot{N}_{\rm fuel}$  = 0.01333 kmol/h

The molar and mass flow rates of the liquid water are

$$\dot{N}_{\rm w} = N_{\rm w} \dot{N}_{\rm fuel} = (1.503 \text{ kmol/kmol fuel})(0.01333 \text{ kmol fuel/h}) = 0.02003 \text{ kmol/h}$$

$$\dot{m}_{\rm w} = \dot{N}_{\rm w} M_{\rm w} = (0.02003 \text{ kmol/h})(18 \text{ kg/kmol}) = 0.3608 \text{ kg/h}$$

The volume flow rate of liquid water is

$$\dot{\mathbf{V}}_{w} = (\mathbf{v}_{f@25^{\circ}C})\dot{m}_{w} = (0.001003 \text{ m}^{3}/\text{kg})(0.3608 \text{ kg/h}) = 0.0003619 \text{ m}^{3}/\text{h} = \mathbf{8.7 L/day}$$

**15-80** Propane is burned with stoichiometric and 50 percent excess air. The adiabatic flame temperature is to be determined for both cases.

Assumptions 1 Steady operating conditions exist. 2 Air and combustion gases are ideal gases. 3 Kinetic and potential energies are negligible. 4 There are no work interactions. 5 The combustion chamber is adiabatic.

Analysis Under steady-flow conditions the energy balance  $E_{in} - E_{out} = \Delta E_{system}$  applied on the combustion chamber with Q = W = 0 reduces to

$$\sum N_P \left( \overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ} \right)_P = \sum N_R \left( \overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ} \right)_R \longrightarrow \sum N_P \left( \overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ} \right)_P = \sum N_R \overline{h}_{f,R}^{\,\circ}$$

since all the reactants are at the standard reference temperature of 25°C. Then, for the stoicihiometric air

$$C_3H_8 + a_{th}(O_2 + 3.76 N_2) \longrightarrow 3 CO_2 + 4 H_2O + a_{th} \times 3.76 N_2$$

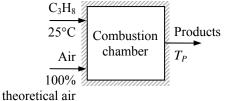
where  $a_{th}$  is the stoichiometric coefficient and is determined from the  $O_2$  balance,

$$a_{th} = 3 + 2 = 5$$

Thus,

$$C_3H_8 + 5(O_2 + 3.76 N_2) \longrightarrow 3CO_2 + 4H_2O + 18.8 N_2$$

From the tables,



	$\overline{\mathbf{h}}_{\mathbf{f}}^{\circ}$	$\overline{\mathbf{h}}_{\mathbf{298K}}$
Substance	kJ/kmol	kJ/kmol
$C_3H_8(g)$	-103,850	
$\mathrm{O}_2$	0	8682
$N_2$	0	8669
$H_2O(g)$	-241,820	9904
$CO_2$	-393,520	9364

Thus,

$$(3)\left(-393,520 + \overline{h}_{CO2} - 9364\right) + (4)\left(-241,820 + \overline{h}_{H2O} - 9904\right) + (18.8)\left(0 + \overline{h}_{N2} - 8669\right) = (1)\left(-103,850\right) + 0 + 0$$
It yields 
$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 18.8\overline{h}_{N2} = 2,274,680 \text{ kJ}$$

The adiabatic flame temperature is obtained from a trial and error solution. A first guess is obtained by dividing the right-hand side of the equation by the total number of moles, which yields 2,274,680/(3+4+18.8)=88,166 kJ/kmol. This enthalpy value corresponds to about 2650 K for N<sub>2</sub>. Noting that the majority of the moles are N<sub>2</sub>,  $T_P$  will be close to 2650 K, but somewhat under it because of the higher specific heat of H<sub>2</sub>O.

At 2500 K:

$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 18.8\overline{h}_{N2} = 3 \times 131,290 + 4 \times 108,868 + 18.8 \times 82,981$$
  
= 2,389,380 kJ (Higher than 2,274,680 kJ)

At 2450 K:

$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 18.8\overline{h}_{N2} = 3 \times 128,219 + 4 \times 106,183 + 18.8 \times 81,149$$
  
= 2,334,990 kJ (Higher than 2,274,680 kJ)

At 2400 K:

$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 18.8\overline{h}_{N2} = 3 \times 125,152 + 4 \times 103,508 + 18.8 \times 79,320$$
  
= 2,280,704 kJ (Higher than 2,274,680 kJ)

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At 2350 K:

$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 18.8\overline{h}_{N2} = 3 \times 122,091 + 4 \times 100,846 + 18.8 \times 77,496$$
  
= 2,226,580 kJ (Lower than 2,274,680 kJ)

By interpolation of the two results,

$$T_P = 2394 \text{ K} = 2121^{\circ}\text{C}$$

When propane is burned with 50% excess air, the reaction equation may be written as

$$C_3H_8 + 1.5 \times a_{th}(O_2 + 3.76N_2) \longrightarrow 3CO_2 + 4H_2O + 0.5 \times a_{th}O_2 + 1.5 \times a_{th} \times 3.76N_2$$

where  $a_{th}$  is the stoichiometric coefficient and is determined from the  $O_2$  balance,

$$1.5a_{th} = 3 + 2 + 0.5a_{th} \longrightarrow a_{th} = 5$$
Thus,
$$C_{3}H_{8} + 7.5(O_{2} + 3.76N_{2}) \longrightarrow 3CO_{2} + 4H_{2}O + 2.5O_{2} + 28.2N_{2}$$

$$C_{3}H_{8} + 7.5(O_{2} + 3.76N_{2}) \longrightarrow 3CO_{2} + 4H_{2}O + 2.5O_{2} + 28.2N_{2}$$

$$C_{3}H_{8} + 7.5(O_{2} + 3.76N_{2}) \longrightarrow 3CO_{2} + 4H_{2}O + 2.5O_{2} + 28.2N_{2}$$

$$C_{3}H_{8} + 7.5(O_{2} + 3.76N_{2}) \longrightarrow 3CO_{2} + 4H_{2}O + 2.5O_{2} + 28.2N_{2}$$

$$C_{3}H_{8} + 7.5(O_{2} + 3.76N_{2}) \longrightarrow 3CO_{2} + 4H_{2}O + 2.5O_{2} + 28.2N_{2}$$

Using the values in the table.

$$(3) \left(-393,520 + \overline{h}_{\text{CO2}} - 9364\right) + (4) \left(-241,820 + \overline{h}_{\text{H2O}} - 9904\right) + (2.5) \left(0 + \overline{h}_{\text{O2}} - 8682\right) + (28.2) \left(0 + \overline{h}_{\text{N2}} - 8669\right) = (1) \left(-103,850\right) + 0 + 0$$

It yields 
$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 2.5\overline{h}_{O2} + 28.2\overline{h}_{N2} = 2,377,870 \text{ kJ}$$

The adiabatic flame temperature is obtained from a trial and error solution. A first guess is obtained by dividing the right-hand side of the equation by the total number of moles, which yields 2,377,870/(3+4+2.5+28.2) = 63,073 kJ/kmol. This enthalpy value corresponds to about 1960 K for N<sub>2</sub>. Noting that the majority of the moles are N<sub>2</sub>,  $T_P$  will be close to 1960 K, but somewhat under it because of the higher specific heat of H<sub>2</sub>O.

At 1800 K:

$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 2.5\overline{h}_{O2} + 28.2\overline{h}_{N2} = 3 \times 88,806 + 4 \times 72,513 + 2.5 \times 60,371 + 28.2 \times 57,651$$
  
= 2.333,160 kJ (Lower than 2.377,870 kJ)

At 1840 K:

$$3\overline{h}_{CO2} + 4\overline{h}_{H2O} + 2.5\overline{h}_{O2} + 28.2\overline{h}_{N2} = 3 \times 91,196 + 4 \times 74,506 + 2.5 \times 61,866 + 28.2 \times 59,075$$
  
= 2,392,190 kJ (Higher than 2,377,870 kJ)

By interpolation,

$$T_P = 1830 \text{ K} = 1557^{\circ}\text{C}$$

**15-89** Ethylene gas is burned steadily with 20 percent excess air. The temperature of products, the entropy generation, and the exergy destruction (or irreversibility) are to be determined.

Assumptions 1 Combustion is complete. 2 Steady operating conditions exist. 3 Air and the combustion gases are ideal gases. 4 Changes in kinetic and potential energies are negligible.

*Analysis* (a) The fuel is burned completely with the excess air, and thus the products will contain only  $CO_2$ ,  $H_2O$ ,  $N_2$ , and some free  $O_2$ . Considering 1 kmol of  $C_2H_4$ , the combustion equation can be written as

$$C_2H_4(g)+1.2a_{th}(O_2+3.76N_2) \longrightarrow 2CO_2+2H_2O+0.2a_{th}O_2+(1.2)(3.76)a_{th}N_2$$

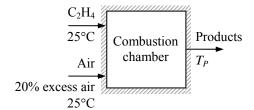
where  $a_{th}$  is the stoichiometric coefficient and is determined from the  $O_2$  balance,

$$1.2a_{th} = 2 + 1 + 0.2a_{th} \longrightarrow a_{th} = 3$$

Thus,

$$C_2H_4(g) + 3.6(O_2 + 3.76N_2) \longrightarrow 2CO_2 + 2H_2O + 0.6O_2 + 13.54N_2$$

Under steady-flow conditions, the exit temperature of the product gases can be determined from the steady-flow energy equation, which reduces to



$$\sum N_P \left(\overline{h}_f^{\circ} + \overline{h} - \overline{h}^{\circ}\right)_P = \sum N_R \overline{h}_{f,R}^{\circ} = \left(N\overline{h}_f^{\circ}\right)_{C_2H_4}$$

since all the reactants are at the standard reference state, and for O2 and N2. From the tables,

	$\overline{\mathbf{h}}_{\mathbf{f}}^{\circ}$	$\overline{\mathbf{h}}_{\mathbf{298\ K}}$	
Substance	kJ/kmol	kJ/kmol	
$C_2H_4(g)$	52,280		
$O_2$	0	8682	
$N_2$	0	8669	
$H_2O(g)$	-241,820	9904	
$CO_2$	-393,520	9364	

Substituting,

or,

$$(2)(-393,520 + \overline{h}_{CO_2} - 9364) + (2)(-241,820 + \overline{h}_{H_2O} - 9904) + (0.6)(0 + \overline{h}_{O_2} - 8682) + (13.54)(0 + \overline{h}_{N_2} - 8669) = (1)(52,280) 2\overline{h}_{CO_2} + 2\overline{h}_{H_2O} + 0.6\overline{h}_{O_2} + 13.54\overline{h}_{N_2} = 1,484,083 \text{ kJ}$$

By trial and error,

$$T_P = 2269.6 \text{ K}$$

(b) The entropy generation during this adiabatic process is determined from

$$S_{\rm gen} = S_P - S_R = \sum N_P \overline{s}_P - \sum N_R \overline{s}_R$$

The  $C_2H_4$  is at 25°C and 1 atm, and thus its absolute entropy is 219.83 kJ/kmol·K (Table A-26). The entropy values listed in the ideal gas tables are for 1 atm pressure. Both the air and the product gases are at a total pressure of 1 atm, but the entropies are to be calculated at the partial pressure of the components which is equal to  $P_i = y_i P_{\text{total}}$ , where  $y_i$  is the mole fraction of component i. Also,

$$S_i = N_i \overline{s}_i(T, P_i) = N_i \left( \overline{s}_i^{\circ}(T, P_0) - R_u \ln(y_i P_m) \right)$$

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The entropy calculations can be presented in tabular form as

	$N_i$	$\mathbf{y_i}$	$ar{\mathbf{s}}_{\mathbf{i}}^{\circ}ig(\mathbf{T,1atm}ig)$	$R_u ln(y_i P_m)$	$N_i \bar{s}_i$
C <sub>2</sub> H <sub>4</sub>	1	1.00	219.83		219.83
$O_2$	3.6	0.21	205.14	-12.98	784.87
$N_2$	13.54	0.79	191.61	-1.96	2620.94
				,	$S_R = 3625.64 \text{ kJ/K}$
CO <sub>2</sub>	2	0.1103	316.881	-18.329	670.42
$H_2O$	2	0.1103	271.134	-18.329	578.93
$O_2$	0.6	0.0331	273.467	-28.336	181.08
$N_2$	13.54	0.7464	256.541	-2.432	3506.49
					$S_P = 4936.92 \text{ kJ/K}$

Thus,

$$S_{\text{gen}} = S_P - S_R = 4936.92 - 3625.64 =$$
**1311.28 kJ/kmol·K**

and

(c) 
$$X_{\text{destroyed}} = T_0 S_{\text{gen}} = (298 \text{ K})(1311.28 \text{ kJ/kmol·K } C_2 H_4) = 390,760 \text{ kJ } (\text{per kmol } C_2 H_4)$$

**15-95** Methyl alcohol is burned steadily with 200 percent excess air in an automobile engine. The maximum amount of work that can be produced by this engine is to be determined.

Assumptions 1 Combustion is complete. 2 Steady operating conditions exist. 3 Air and the combustion gases are ideal gases. 4 Changes in kinetic and potential energies are negligible.

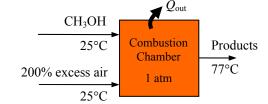
Analysis The fuel is burned completely with the excess air, and thus the products will contain only  $CO_2$ ,  $H_2O$ ,  $N_2$ , and some free  $O_2$ . Considering 1 kmol  $CH_3OH$  the combustion equation can be written as

$$CH_3OH + 3a_{th}(O_2 + 3.76N_2) \longrightarrow CO_2 + 2H_2O + 2a_{th}O_2 + 3a_{th} \times 3.76N_2$$

where  $a_{th}$  is the stoichiometric coefficient and is determined from the  $O_2$  balance,

$$0.5 + 3a_{th} = 1 + 1 + 2a_{th} \longrightarrow a_{th} = 1.5$$

Thus,



$$CH_3OH + 4.5(O_2 + 3.76N_2) \longrightarrow CO_2 + 2H_2O + 3O_2 + 16.92N_2$$

Under steady-flow conditions the energy balance  $E_{\rm in} - E_{\rm out} = \Delta E_{\rm system}$  applied on the combustion chamber with W = 0 reduces to

$$-Q_{\mathrm{out}} = \sum N_P \Big(\overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ}\Big)_{\!P} - \sum N_R \Big(\overline{h}_f^{\,\circ} + \overline{h} - \overline{h}^{\,\circ}\Big)_{\!R}$$

Assuming the air and the combustion products to be ideal gases, we have h = h(T). From the tables,

	$\overline{\mathbf{h}}_{\mathbf{f}}^{\circ}$	$\overline{\mathbf{h}}_{\mathbf{298K}}$	$\overline{\mathbf{h}}_{350\mathbf{K}}$
Substance	kJ/kmol	kJ/kmol	kJ/kmol
CH <sub>3</sub> OH	-200,670		
$O_2$	0	8682	10,213
$N_2$	0	8669	10,180
$H_2O(g)$	-241,820	9904	11,652
$CO_2$	-393,520	9364	11,351

Thus.

$$-Q_{\text{out}} = (1)(-393,520 + 11,351 - 9364) + (2)(-241,820 + 11,652 - 9904) + (3)(0 + 10,213 - 8682) + (16.92)(0 + 10,180 - 8669) - (1)(-200,670) = -663.550 \text{ kJ/kmol of fuel}$$

The entropy generation during this process is determined from

$$S_{\text{gen}} = S_P - S_R + \frac{Q_{\text{out}}}{T_{\text{surr}}} = \sum N_P \overline{s}_P - \sum N_R \overline{s}_R + \frac{Q_{\text{out}}}{T_{\text{surr}}}$$

The entropy values listed in the ideal gas tables are for 1 atm pressure. Both the air and the product gases are at a total pressure of 1 atm, but the entropies are to be calculated at the partial pressure of the components which is equal to  $P_i = y_i$   $P_{\text{total}}$ , where  $y_i$  is the mole fraction of component i. Then,

$$S_i = N_i \overline{s}_i (T, P_i) = N_i (\overline{s}_i^{\circ} (T, P_0) - R_u \ln(y_i P_m))$$

The entropy calculations can be presented in tabular form as

	$N_i$	$\mathbf{y_i}$	$ar{\mathbf{s}}_{\mathbf{i}}^{\circ} ig(\mathbf{T,1atm}ig)$	$R_u ln(y_i P_m)$	$N_i \bar{s}_i$
CH <sub>3</sub> OH	1		239.70		239.70
$O_2$	4.5	0.21	205.04	-12.98	981.09
$N_2$	16.92	0.79	191.61	-1.960	3275.20
					$S_R = 4496 \text{ kJ/K}$
CO <sub>2</sub>	1	0.0436	219.831	-26.05	245.88
$H_2O(g)$	2	0.0873	194.125	-20.27	428.79
$O_2$	3	0.1309	209.765	-16.91	680.03
$N_2$	16.92	0.7382	196.173	-2.52	3361.89
					$S_P = 4717 \text{ kJ/K}$

Thus,

$$S_{\text{gen}} = S_P - S_R + \frac{Q_{\text{out}}}{T_{\text{surr}}} = 4717 - 4496 + \frac{663,550}{298} = 2448 \text{ kJ/K (per kmol fuel)}$$

The maximum work is equal to the exergy destruction

$$W_{\text{max}} = X_{\text{dest}} = T_0 S_{\text{gen}} = (298)(2448 \text{ kJ/K}) = 729,400 \text{ kJ/K} \text{ (per kmol fuel)}$$

Per unit mass basis,

$$W_{\rm max} = \frac{729,400 \, {\rm kJ/K \cdot kmol}}{32 \, {\rm kg/kmol}} =$$
 22,794 kJ/kg fuel